



*(Knowledge for Development)*

**KIBABII UNIVERSITY**  
**UNIVERSITY EXAMINATIONS**  
**2017/2018 ACADEMIC YEAR**  
**FOURTH YEAR FIRST SEMESTER**  
**SPECIAL/SUPPLEMENTARY EXAMINATION**  
**FOR THE DEGREE OF BACHELOR OF SCIENCE**  
**(MATHEMATICS)**

**COURSE CODE:** MAT 425

**COURSE TITLE:** FLUID MECHANICS II

**DATE:** 05/10/18

**TIME:** 11.30 AM -1.30 PM

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**INSTRUCTIONS TO CANDIDATES**

Answer Question One and Any other TWO Questions

TIME: 2 Hours

This Paper Consists of 4 Printed Pages. Please Turn Over.

**QUESTION ONE (30 MARKS)**

- a. Find the equations of the streamlines due to uniform line source of strength  $m$  through the points A(-c, 0), B(c, 0) and a uniform line sink of strength  $2m$  through the origin. (6 marks)
- b. The velocity potential function for a two dimensional flow is  $\Phi = x(2y - 1)$ . At a point P(4, 5) determine;
- i) The velocity (3 marks)
- ii) The value of the stream function (4 marks)
- c. The resistance force R of a supersonic plane during flight can be considered as dependent upon the length of the aircraft l, velocity v, air viscosity  $\mu$ , air density  $\rho$  and bulk modulus of air k. Find an expression for R. (6 marks)
- d. Show that for an incompressible steady flow with constant velocity components

$$u(y) = y \frac{U}{h} + \frac{h^2}{2\mu} \left( -\frac{dp}{dx} \right) \frac{y}{h} \left( 1 - \frac{y}{h} \right)$$
$$v = w = 0$$

Satisfy the equation of motion, when the body force is neglected  $h, U, \frac{dp}{dx}$  are constants and  $P = P(x)$ . (6 marks)

- e. Show that for a two dimensionally axially symmetric boundary layer flow.

$$\int_0^\infty \left( 1 - \frac{u}{U} \right)^2 \frac{r}{\alpha} dn = \delta_1 - \delta_2, \int_0^\infty \left( 1 - \frac{u}{U} \right)^3 \frac{r}{\alpha} dn = \delta_1 - 3\delta_2 + \delta_3$$

Where  $n$  the normal distance from the surface of the body is,  $r$  is the axial distance and  $\alpha$  is the reference radius which maybe a function of the axial distance. (5 marks)

**QUESTION TWO (20 MARKS)**

- a. State the Buckingham's  $\pi$ -theorem. (2 marks)
- b. Differentiate between the following terms as used in dimensional analysis (6 marks)
- i) Geometrical symmetry
- ii) Kinematic similarity
- iii) Dynamic similarity
- c. The pressure difference  $\Delta P$  in a pipe of diameter D and length l due to turbulent flow depends on the velocity V, viscosity  $\mu$ , density  $\rho$ , roughness K. Using Buckingham's  $\pi$ -theorem, obtain an expression for  $\Delta P$ . (7 marks)

- d. Using Rayleigh's technique, show that the resistance ( $R$ ) to the motion of a sphere of diameter ( $D$ ) moving with a uniform velocity ( $V$ ) through a real fluid having density  $\rho$  and viscosity  $\mu$  is given by  $R = \rho D^2 V^2 f\left(\frac{\mu}{\rho V D}\right)$ . (5 marks)

**QUESTION THREE (20 MARKS)**

- a. Define the following terms
- i. Sink
  - ii. Source
  - iii. Doublet (3 marks)
- b. Show that the velocity vector  $\mathbf{q}$  is everywhere tangent to the line in the  $xy$  plane along which  $\psi(x, y) = \text{constant}$ . (3 marks)
- c. Discuss the flow for which  $W = z^2$ . (5 marks)
- d. Determine the stream function  $\psi(x, y, t)$  for the given velocity field given by  $u = Ut, v = x$ . (4 marks)
- e. Discuss the flow due to a uniform line doublet at 0 of strength  $\mu$  per unit length and its axis being along OX. (5 marks)

**QUESTION FOUR (20 MARKS)**

- a) State Milne Thomson's circle theorem. (2 marks)
- b) A source and a sink of equal strength are placed at the points  $\left(\pm \frac{1}{2}a, 0\right)$  within a fixed circular boundary  $y^2 + x^2 = a^2$ . Show that the streamlines are given by  $(r^2 - \frac{1}{4}a^2)(r^2 - 4a^2) - 4a^2y^2 = ky(r^2 - a^2)$ . (5 marks)
- c) In the region bounded by a fixed quadrantal arc and its radii, deduce the motion due to a source and an equal sink situated at the ends of one of the bounding radii. Show that the stream line leaving either end at an angle  $\alpha$  with the radius is  $r^2 \sin(\alpha + \theta) = x^2 \sin(\alpha - \theta)$ . (6 marks)
- d) A vortex of circulation  $2\pi k$  is at the point  $z = na$  ( $n > 1$ ) in the presence of a plane circular boundary  $Z = a$ , around which there is a circulation  $2\pi \lambda k$ . Show that

$$\lambda = \frac{1}{(n^2 - 1)} \quad (6 \text{ marks})$$

**QUESTION FIVE (20 MARKS)**

- a) State the theorem of Blasius. (2 marks)
- b) Verify that  $W = iK \log \left\{ \frac{z-ia}{z+ia} \right\}$  is the complex potential of a steady flow of liquid about a cylinder the plane  $y = 0$  being a rigid boundary. Find the forces exerted by the liquid on unit length of the cylinder. (7 marks)
- c) Find the image of a line source in a circular cylinder. (5 marks)
- d) Show that the velocity potential  $\phi = \frac{a}{2} (x^2 + y^2 - 2z^2)$  satisfies the Laplace equation and determine its streamlines. (6 marks)